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THE REMOTE LINK UNIT: A DEMONSTRATION OF OPERATIONAL
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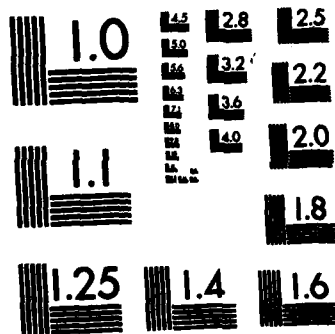
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PART I

THE REMOTE LINK UNIT: A DEMONSTRATION OF OPERATIONAL PERFORMANCE

Part I - Summary

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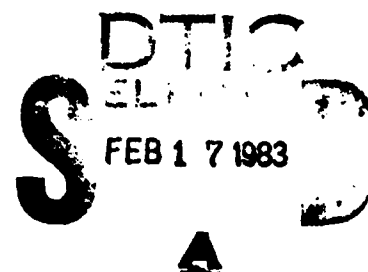
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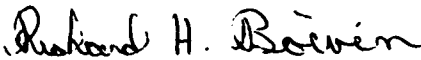


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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) This report summarizes the results of the program of implementation of an RLU Demonstration System. This system allows the evaluation of the most unique parts of the RLU and has facilitated the resolution of problem areas in the RLU design. Enhancements to the RLU functional design are presented and an implementation plan for an RLU prototype is proposed. The User's Manual and the Design Manual for the system are included as Parts II and III, respectively.		



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PREFACE

This document was prepared by the University of Houston, Houston, Texas, on Air Force Contract No. F33615-80-C-1095, entitled "The Remote Link Unit - A Demonstration of Operational Performance."

The work was administered under the direction of the Information Transfer Group, Information Processing Technology Branch, System Avionics Division of the Avionics Laboratory, under Project 2003, "Avionic System Design Technology," Task 08, "Multiplex and Information Transfer Technology," Work Unit 07, "Remote Link Unit Demonstration." The work was performed during the period 1 April 1980 to 31 December 1980 and this report was submitted in August 1981. The Air Force Project Engineer was Philip C. Goldman (AFWAL/AAAT-3).

The work is a continuation of a previous feasibility study entitled, "The Remote Link Unit: An Advanced Remote Terminal for MIL-STD-1553A." The results of this study are documented in a technical report entitled, "Remote Link Unit Functional Design: An Advanced Remote Terminal for MIL-STD-1553B," which was published as AFAL-TR-79-1176, AD-A080126. An add-on to this previous study resulted in a second technical report entitled, "The Remote Link Unit: Applications to the Design for Repair Methodology Program," published as AFWAL-TR-80-1033, AD-A086126.

This report summarizes the design, development, and testing accomplished under the contracted work. The Principal Investigator and Program Manager was Dr. Carlos J. Tavora. Drs. John Glover, Jr. and Miles A. Smither were Co-Investigators. Dr. Tavora was responsible for the system architecture and modularization of the design. Dr. Glover supervised the design of the software for the Link Manager Simulator and the Link Module. He was assisted by Messrs. Hao-Cheng Hsia, William C. Law, and Parmanand Balsaver. Dr. Smither was assisted by Mr. Tzer-Tsan Lin in the design of the Interface Configuration Adapter. Mr. H. Mitchell Collins was in charge of the design of the Electronic Nameplate and the Nameplate Interface Controller.

This report is organized in three parts: Part I - Summary, Part II - User's Manual, and Part III - Design Manual. Part III is separated into two volumes: Volume 1 is the main body of the Design Manual, while Volume 2 contains the appendices.

This part is organized as a short summary of the Remote Link Unit Demonstration project, with an overview, and including design enhancements and an implementation plan for RLU development. A brief summary of test results is also included.

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LIST OF ACRONYMS

AC	Alternating Current
A/D	Analog to Digital
BCD	Binary Coded Decimal
CMOS	Complementary Metal Oxide Semiconductor
CPU	Central Processing Unit
CRT	Cathode Ray Tube
D/A	Digital to Analog
DC	Direct Current
DIP	Dual In-line Package
DMA	Direct Memory Access
EAROM	Electrically Alterable ROM
EEPROM	Electrically Erasable Programmable ROM
EPROM	Erasable Programmable ROM
FP	Front Panel
ICA	Interface Configuration Adapter
I/O	Input/Output
ISR	Interrupt Service Routine
LA	Link Address
LM	Link Module
LMG	Link Manager
LMP	LM Processor
LSB	Least Significant Bit
MP	Maintenance Port
MSB	Most Significant Bit
MUX	Multiplexer
NIC	Nameplate Interface Controller
NP	Nameplate
PCB	Printed Circuit Board
PIA	Peripheral Interface Adapter
PROM	Programmable Read Only Memory
RAM	Random Access Memory
RLU	Remote Link Unit
RLUDS	Remote Link Unit Demonstration System
RMW	Read-Modify-Write
ROM	Read Only Memory
RT	Remote Terminal
SIC	Subsystem Information Channel
SM	Shared Memory
SRU	Shop Replaceable Unit
TTL	Transistor-Transistor Logic
UFT	User File Table

SECTION 1

INTRODUCTION

This report describes the results of a program of implementation of a RLU Demonstration System (RLUDS). The Remote Link Unit (RLU) is a new design concept for remote terminals which is based on the remote terminal developed under the Avionics Laboratory's Digital Avionics Information System (DAIS) program. Design of the RLUDS is based on the functional design described in the document "Remote Link Unit Functional Design: An Advanced Remote Terminal for MIL-STD-1553B," [1]. The RLUDS is not intended to be a complete RLU prototype but it demonstrates the most unique parts of the RLU. The RLUDS is an experimental device that provides operational performance data on the RLU and has facilitated the resolution of problem areas in the RLU design.

1.1 BACKGROUND

Remote terminals (RTs) are used extensively in the current military aviation electronics area. Typically, an aircraft will contain a central computer and several remotely located avionic subsystems (radar, radio, navigation set, controls and displays, etc.). The communication between the central computer and the subsystems is effected through a time-division multiplex data bus, such as MIL-STD-1553B. Remote subsystems are connected to the bus through remote terminals, which have several functions:

- Sending and receiving messages of data and commands over the bus in the proper standard bus protocol,
- Responding to commands from the central computer,
- Sending and receiving signals to the subsystem in the proper format for that particular subsystem, and
- Mapping of subsystem addresses to the particular data channels for the various subsystems.

The Remote Link Unit (RLU) is comparable to the DAIS remote terminal upon which it is based, but it is not limited to being a replacement of the DAIS terminal. It is a collection of diverse and interacting logical and processing functions that impact all types of avionic data communication. The RLU can be described in terms of three major components: the Electronic Nameplate, the Link Module, and the Link Manager.

1.1.1 Electronic Nameplate

The electronic nameplate (NP) is a non-volatile and electrically alterable digital memory that can store information during a mission and save it until it is needed. It also has non-alterable memory, as required. The electronic nameplate is a part of the RLU but is physically attached to the subsystem. It contains information that can be read on demand, such as:

- Subsystem identification (type, model, serial number, etc.),
- Subsystem function,
- Physical location of subsystem,
- Interface specification (level, timing, handshaking, rate, etc.),
- Operational history (calibration, maintenance, etc.), and
- Configuration programs.

The configuration programs are loaded into the RLU's processor to enable it to communicate with the subsystem. These can include initialization, data conversion, calibration, and diagnostic programs.

1.1.2 Link Module

The Link Module (LM) can replace 15 of the 17 different interface modules of the DAIS RT. The Link Module is a standard hardware unit that contains a processor with three interfaces: the Interface Configuration Adapter (ICA), the Nameplate Interface Controller (NIC), and the Shared Memory (SM).

The Interface Configuration Adapter is the device that implements the "universal interface" of the RLU. Under software control, it configures itself to the proper interface specification for communication with the subsystem.

The Nameplate Interface Controller provides RLU access to the electronic nameplate attached to a subsystem. Through the NIC the LM processor retrieves identification, configuration parameters and data conversion programs related to the subsystem.

The Shared Memory implements the interface between the Link Module and the Link Manager. It provides for the storage of commands, status, data and handshake flags.

The processor supports the overall operation of the LM performing the processing of programs that support the operation of the subsystem and the LM interfaces.

1.1.3 Link Manager

The Link Manager (LMG) is the supervisor of the RLU. The LMG contains processing and memory and supports the overall operation of the RLU, including communication; resource control; redundancy management (of dual buses or dual power supplies, for example); fault detection, isolation, and recording; and data conversion. In addition, it has the potential capability to control the RLU and its subsystems in a stand-alone fashion, in the event of loss of communication with the central computer, due to hardware/software failure or perhaps even battle damage. This clustering of functions could greatly enhance the survivability of digital avionic systems in future aircraft.

1.2 OBJECTIVES AND APPROACH

The RLU Demonstration System (RLUDS) was built to achieve three objectives:

- Demonstrate the RLU concepts in operation,
- Support the design of an RLU prototype, and
- Design the SIC and the ICA as stand-alone avionic system components.

Design of an RLU prototype requires that the feasibility of implementation of RLU concepts be demonstrated beyond a functional design. Some of the RLU concepts require hardware and software designs which are not conventional. Construction and operation of the demonstration system was essential to the identification and solution of problems not uncovered in the functional design.

1.2.1 Approach

The RLUDS implementation of the RLU concentrates on the Link Module and its interfaces as depicted in Figure 1. The following RLU components required special design and fabrication:

- Link Module (LM),
- Subsystem Information Channel (SIC) and Electronic Nameplate (NP),
- Interface Configuration Adapter (ICA), and
- Shared Memory (SM) interface between the LM and the Link Manager (LMG).

The Link Manager (LMG) was simulated with a PDP-11/70, while the LM was implemented with a Motorola 6800 microprocessor and the NP utilized a Motorola 6801 microprocessor. The architecture of the RLUDS is shown in Figure 2.

1.3 DELIVERABLES

The RLU Demonstration System consists of hardware, software and documents as described below.

1.3.1 Hardware

The hardware of the RLUDS includes the following items:

- One Link Module with processor, memory, three interface cards and power supply. The link module is housed in a two-chassis enclosure equipped with connectors and a front panel display module.

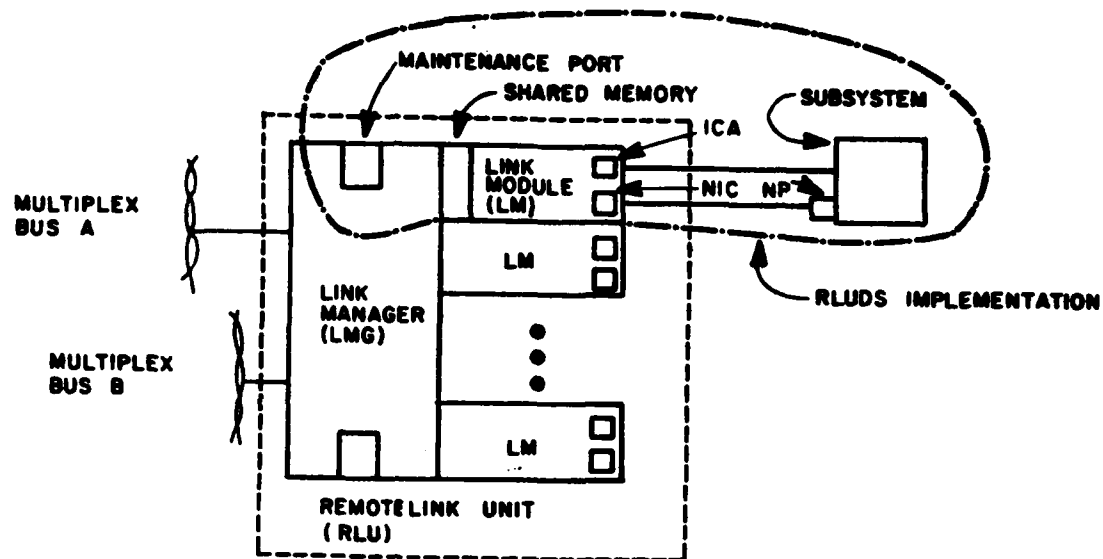


Figure 1 RLUDS Implementation of the RLU

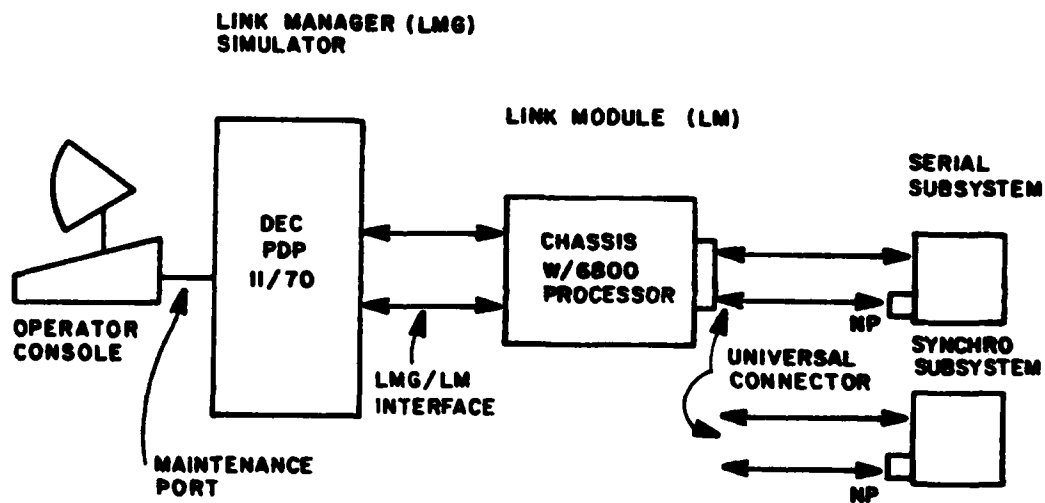


Figure 2 RLUDS Configuration

- Two electronic nameplates that interface to the LM through a Subsystem Information Channel (SIC).
- Two distinct subsystems that can be used to demonstrate the processing and interfacing capabilities of the RLU. One subsystem contains two synchros which are used as input and output devices. The other subsystem requires a serial digital interface to support switches as input devices and a four digit display as an output device. The serial input can operate in either flag or refresh mode.
- A Test box that exercises the ICA by generating and detecting signals corresponding to 15 different interface modules utilized by DAIS.
- A set of cables to interface the LM with its subsystems and the PDP-11/70 LMG simulator.

1.3.2 Software

Two tapes containing RLUDS software are delivered with the system. Both tapes are recorded on 9 tracks with 800 bpi density using DOS (DEC) format. The first tape - labeled "RLUDS Source Software" - contains software for the Link Manager simulator, the Link Module and the Electronic Nameplate. The second tape - labeled "RLUDS Installation Tape" - is used to install the Link Manager software on a PDP-11/70 series host computer.

1.3.3 Documentation

Three documents are part of the RLU demonstration system. These documents are as follows:

1.3.3.1 User's Manual

Part II of this document contains the User's Manual for the RLU Demonstration System. This manual describes how the system should be set up for operation, and presents a tutorial on the use of the LMG man-machine interactive dialogue program. Utilization of the LM's front panel controls and displays is also included. Test scripts designed to exercise all RLUDS functions are also included.

1.3.3.2 Design Manual

Part III of this document contains the Design Manual for the RLU Demonstration System. This manual presents a detailed description of the hardware and software utilized in the implementation of the system. The theory of operation of all system modules is presented and the interaction between hardware and software is described. This manual is in two volumes. Volume 1 is the basic Design Manual and Volume 2 is the Appendices A, B, and C.

1.3.3.3 Test Plan

The document "Test and Acceptance Plan for the RLU Demonstration System," [2], outlines a test procedure to evaluate all major RLU Demonstration system components. The test plan incorporates three sets of tests: RLU tests, ICA tests, and SIC tests.

The RLU tests are tests designed to demonstrate the performance of the RLU as a system. In these tests, the Link Manager interacts with the Link Module to control a subsystem. Failures induced in the subsystem demonstrate the fault monitoring and recording capabilities of the RLU.

The ICA tests demonstrate the performance of the ICA as a universal interface module.

The SIC tests demonstrate that the Subsystem Information Channel is a viable distributed memory device that supports the recording of subsystem faults and is compatible with the retrieval of maintenance data at a later time.

SECTION 2

FUNCTIONAL DESIGN ENHANCEMENTS

During the development of the RLUDS, several new features were included which are now proposed as enhancements to the previous RLU design. Changes to the RLU Functional Design [1] necessary to incorporate these enhancements are outlined below.

2.1 LINK MODULE AND LINK MANAGER

The only changes to the Functional Design in the chapters describing the LM and the LMG relate to an enhancement which provides a more flexible message transfer capability. Previously, two data transfer buffers were required in shared memory to allow all data transfers to be double-buffered. Realizing, however, that not all messages possess the real-time urgency requiring double-buffering, the message exchange protocol was enhanced to allow the option of single or double buffers. When double-buffering is selected, data transfer is essentially as described in the Functional Design. When single-buffering is selected, Buffer 0 can be used for output while, concurrently, Buffer 1 is used for input. This option increases the utility of the shared memory buffers.

The effect of this enhancement on the Functional Design is to change those sections which describe the data transfer protocol.

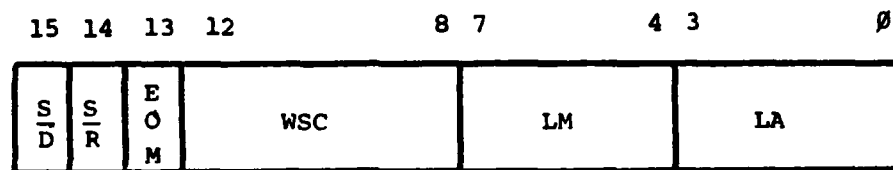
2.1.1 Mapping EAROM Changes

Section 6.1.3 of the Functional Design describes the mapping EAROM in the LMG. This table remains unchanged except for a modification of two bits in each 16-bit entry in the EAROM. The format of each map entry, as illustrated in Figure 37 of the Functional Design, should be modified to be as shown in Figure 3. The S/D bit indicates whether single or double-buffering is selected for this message. The S/R bit indicates whether the message is a sequential or refreshed type of message. The distinction between sequential and refreshed is described below.

2.1.2 Shared Memory Handshake

Sections 3.1.3 and 6.2.4 of the Functional Design describe the handshake in shared memory which implements the data transfers. This handshake is now modified to allow the enhancements described earlier. The new handshaking protocol is described here:

First, two types of I/O are provided for: sequential and refreshed. A sequential message is placed once into the shared memory buffer and must be received and acknowledged to avoid an error condition. In DAIS, all output messages are sequential and all asynchronous input messages are sequential. A refreshed message is one in which the data is simply kept up-to-date in the shared memory buffers and is only occasionally received. A message is overwritten by a subsequent message without causing an error condition. In DAIS, synchronous input messages are refreshed.



S/D = Single-/double-buffered

S/R = Sequential/refreshed

EOM = end of message

WSC = word subcount

LM = link module no.

LA = link address

Figure 3 Format of Mapping EAROM Entry

Second, the data transfer status word has been slightly rearranged to accommodate the new I/O features. Figure 4 illustrates the new handshake protocol and replaces Figure 13 of the Functional Design. After command byte C0 is written to shared memory, the LMG performs a read-modify-write operation on status byte S1. After the read portion, the LMG checks the RDY0 and the RDY1 bits. For single-buffered I/O, RDY0 will be set if the output buffer is available and RDY1 will be set if the input buffer is available. For double-buffered I/O, the RDY0 and RDY1 bits indicate which of the two buffers is available. In either case, if a buffer is available, the LMG sets the REQ0 or REQ1 bit to indicate which buffer is taken and writes the S1 byte back to shared memory to complete the read-modify-write. The LMG then writes the S0 byte to shared memory to indicate the number of words to be involved in the data transfer. Other details of the handshake, such as the use of command bytes C0 and C1, remain unchanged.

2.2 INTERFACE CONFIGURATION ADAPTOR

Development of the detailed design for the Interface Configuration Adaptor led to modifications of the functional design that enhance the operation. These modifications are outlined below.

2.2.1 Signal Interface

In view of the high resistance associated with solid-state switches, it was necessary to provide a direct current path for the single-ended connection. With the new design, the differential output is applied to the subsystem through the two differential drivers of the signal I/O section of the ICA. The single-ended output is available between the positive driver and ground. A diagram depicting the connections for the single-ended and differential outputs is illustrated in Figure 5.

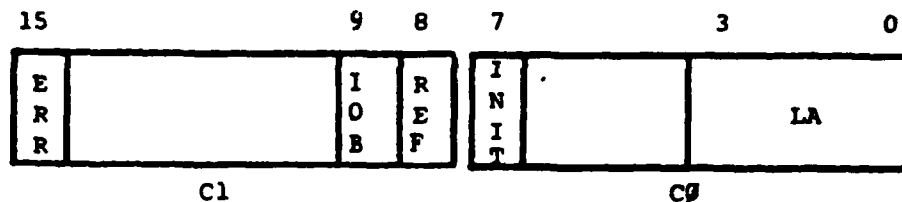
2.2.2 Output Signal Multiplexing

The generation of analog and digital output signals described in the functional design utilizes a digital multiplexer to select digital data corresponding to analog or digital values for the D/A converter. A more effective scheme which reduces the number of lines used was implemented. This design utilizes an analog multiplexer to select analog signals corresponding to the analog or digital value to be output. A diagram illustrating the new design is presented in Figure 6.

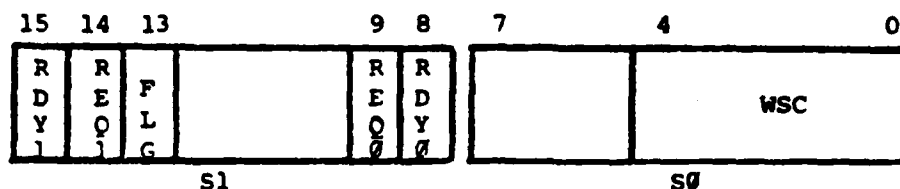
2.2.3 AC Reference Signal

Generation of a 400 Hz AC reference signal was incorporated as a new ICA function. This signal is synthesized from the digital system clock through the use of a counter, a sine table stored in ROM and a D/A converter. This signal can be used by subsystems as an AC reference thus eliminating the need for external reference signals. Generation of the AC reference at the ICA simplifies the synchronization of sampling with AC input signals which must be converted at their peak value.

Data Transfer Command Word:



Data Transfer Status Word:



Data Transfer Handshake:

1. LMG sends C0 to LM, generating an interrupt

LA = 4-bit Link Address

INIT = 1 to indicate initiation of message transfer

2. LMG performs read-modify-write on S1

FLG = 1 if subsystem is flagged as down

RDY0 = 1 if buffer 0 is available

RDY1 = 1 if buffer 1 is available

REQ0 = 1 to indicate buffer 0 is being taken

REQ1 = 1 to indicate buffer 1 is being taken

3. LMG writes S0 to LM (output) or reads S0 from LM (input)

WSC = word subcount, no. of words being transferred

4. LMG transfers data

5. LMG sends C1 to LM, generating an interrupt

ERR = 1 if an error was encountered

IOB = 1 to release buffer 1, 0 to release buffer 0 (single-buffered I/O)

REF = 1 if buffer to be released is refreshed

Figure 4 Data Transfer Handshake Protocol

ICA
INTERFACE

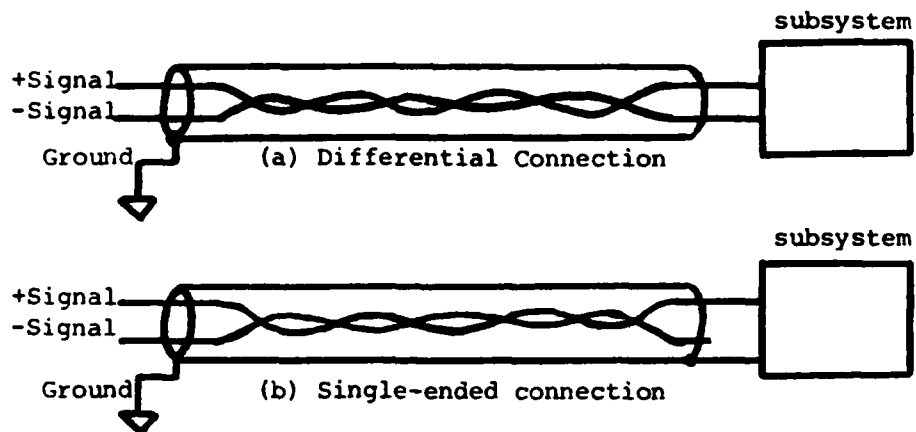


Figure 5 ICA Signal I/O Connections

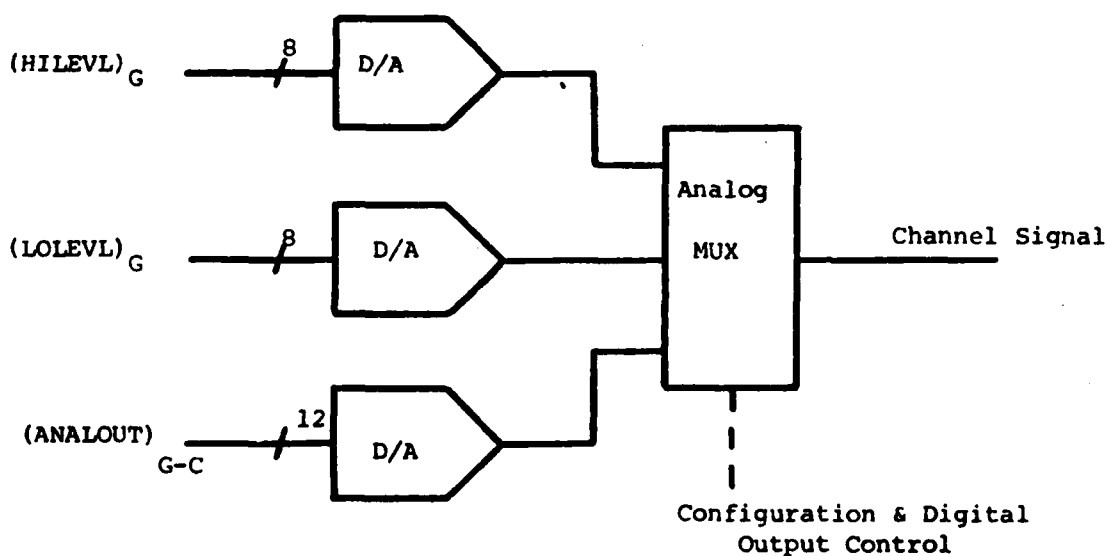


Figure 6 Analog and Digital Output Multiplexing

2.2.4 Synchro Reference

Incorporation of the AC reference as part of the ICA allows a group configured for synchro operation to utilize the fourth channel to provide the AC reference. This signal can be connected to the reference winding of a synchro.

2.2.5 Serial Input Handshake

A modification was introduced in the handshake associated with the flag mode of the serial digital input. A flag mode transmission is initiated by the subsystem by raising the flag/acknowledge line. A handshake is provided by the ICA through the request/lockout line. A modification of the handshake was introduced to handle error occurrences. With this modification, the flag/acknowledge signal remains high when an error occurs during a transmission to the ICA. Upon detecting an error, the ICA lowers the request/lockout signal. This action will not cause the flag/acknowledged signal to go low. The request/lockout signal will remain low for a short period of time to reset the serial subsystem. A new transmission is initiated when the request/lockout signal goes high. A diagram illustrating the handshake modification is presented in Figure 7.

2.2.6 Interface Buffer Design

On the functional design, the read and write interface buffers between the ICA and the LM were given separate address assignments. By overlapping these addresses a reduced address space is used. This feature does not alter the operational capabilities of the ICA.

2.2.7 Output Automatic Scaling

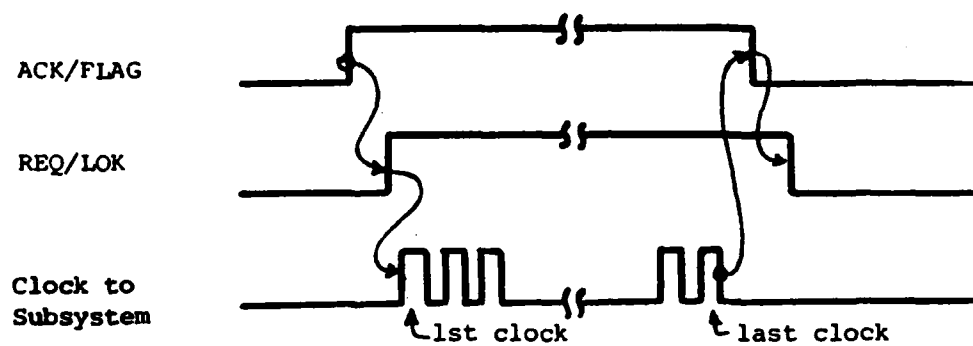
Automatic scaling of the AC and DC reference with selection of single-ended or differential output was introduced. The automatic scaling guarantees that a digital value sent to the D/A will produce an output voltage that is independent of the output configuration used (single-ended or differential).

2.3 SUBSYSTEM INFORMATION CHANNEL

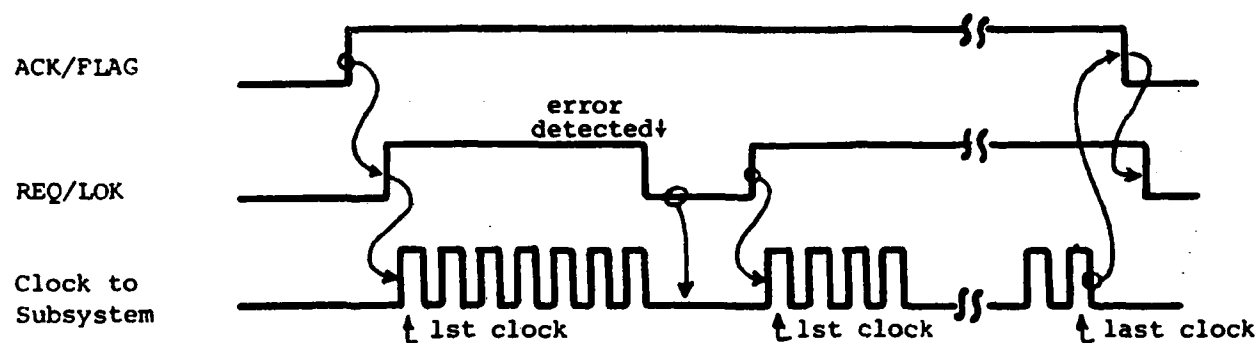
Changes in the Subsystem Information Channel (SIC) all stem from the decision to implement the electronic nameplate (NP) with a microprocessor. Use of a microprocessor increases the flexibility of the NP, but does not greatly change the basic NP functions. Principal changes to the Functional Design are:

2.3.1 Nameplate Clock

The clock signal to the NP is always present to enable operation of the microprocessor. Instead of stopping and starting the clock as described in sections 5.1 and 5.3 of the Functional Design, another line is added to



(a) Transmission with no errors



(b) Transmission with one error detected

Figure 7 Flag Mode Serial Input Handshake

the SIC bus. This line is the RESPONSE/COMMAND line and is used to indicate when a command is being transmitted to the NP and when the NP is expected to respond. The use of this line is illustrated in Figure 8.

2.3.2 Nameplate Commands

Section 5.2 and Table 9 of the Functional Design should be modified to allow commands for transmitting more than one word of data at a time between the LM and the NP. Table 1 provides a more flexible command structure and would replace Table 9 of the Functional Design.

2.3.3 Nameplate Architecture

Because of the microprocessor implementation, Figure 31 and 32 in the functional design do not apply. Instead, the NP architecture simplifies to a single-chip microprocessor, and the state diagram is replaced by a NP software flow chart.

2.3.4 Nameplate Diagnostic

The intelligence in the NP allows a NP self-diagnostic to be implemented by the microprocessor, a capability not provided for in the Functional Design.

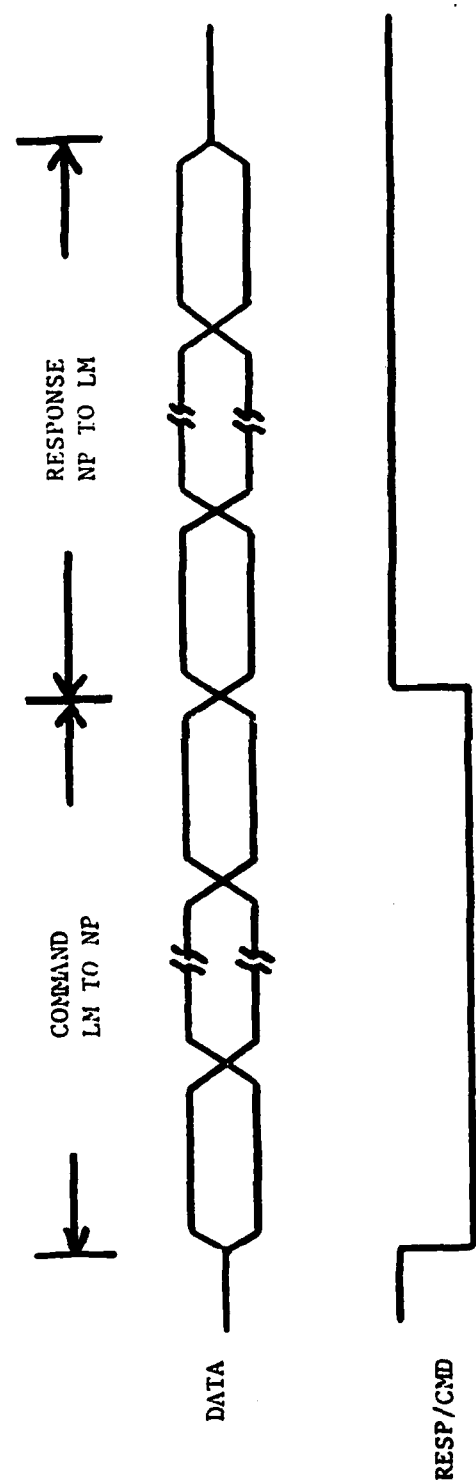


Figure 8 New COMMAND/RESPONSE Protocol

TABLE 1 NAMEPLATE COMMANDS

COMMAND		RESPONSE	
Instruction	Data	Status	Data
1. Select level 0 NP	N/A	NP status	NP address
2. Select next IIP	N/A	NP status	Newly selected NP address
3. Select NP by address	NP address	NP status	NP address
4. Deselect NP	N/A	NP status	NP address
5. Assign NP address	Address to be assigned	NP status	NP address
6. Read selected NP's address	N/A	NP status	NP address
7. Read selected IIP's memory	Starting memory address, Number of bytes wanted	NP status	Starting memory address, memory data
8. Write enable/ disable	Enable/disable flag (Enable =1)	NP status	NP address
9. Write memory data	Starting memory address, Number of bytes, Data bytes	NP status	Starting memory address
10. Next available address to be written	N/A	NP status	Address of next available write memory record
11. Erase write memory	N/A	NP status	NP address
12. Run NP diagnostic	N/A	NP status	NP address
13. Read NP diagnostic results	N/A	NP status	Diagnostic results
14. Reset selected NP	N/A	NP status	NP address

SECTION 3

IMPLEMENTATION PLAN

This section describes a plan for the development of an RLU prototype and its subsequent evolution into a set of distributed components that can be used in future avionic system design. The proposed plan requires two phases for implementation. The first phase will be referred to as the RLU Integration Phase and the second the RLU Distribution Phase.

3.1 OBJECTIVES AND APPROACH

The program of development outlined in this plan will lead to the dissemination of the technology developed for the Remote Link Unit and will facilitate its use in various avionic system designs. The following objectives should be pursued in the development of an RLU prototype:

- The RLU should be usable as a universal maintenance fixture that will support the maintenance of avionic systems.
- The RLU should provide an information interface that simplifies the design of avionics systems and provides a well defined separation between subsystems and system functions.
- The RLU prototype should yield components that simplify the design of subsystems by providing information oriented interfaces that support fault monitoring and recording.

The above objectives can be achieved with a program structured with a two-phase development approach as described in the following sections.

3.1.1 Phase I - RLU Integration

During this phase a prototype RLU would be developed and a program of utilization of the RLU as a maintenance support device would take place.

Development of the RLU prototype should utilize a parallel implementation approach as illustrated in Figure 9. This parallel approach will lead to the development of modules that can be used independently of the RLU. RLUDS design manuals (Part III) should provide a basis for the design of the new modules.

A program of utilization of the RLU as a maintenance console which is configured through the use of electronic nameplates should evolve with the design of the RLU prototype. Diagnostic programs stored in the electronic nameplates of avionic shop replaceable units (SRUs) will be used to configure the RLU to perform its maintenance support function.

During this phase, a study should be conducted to specify the structure and format of the subsystem data (information) stored in the LM's shared memory. This information should be universal in nature, that is, it should express the value of a variable in engineering units and its representation should be in a standard floating point format. The standardization of an

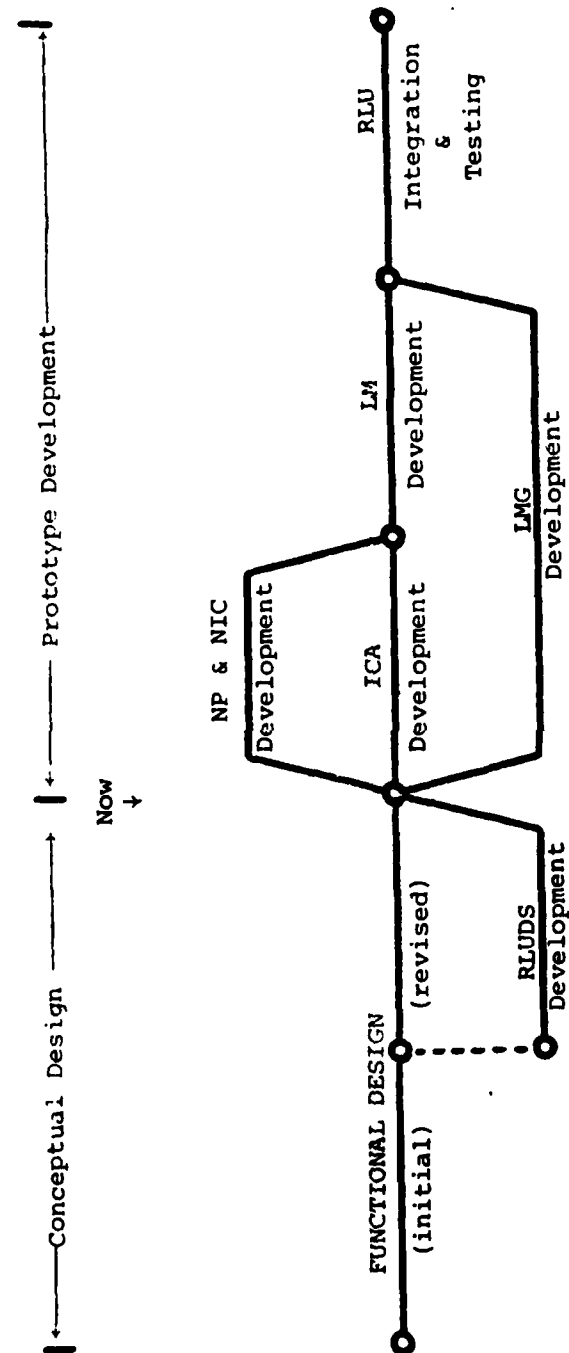


Figure 9 RLU Implementation Chart

information interface will enable the design of system software that is independent of subsystem hardware peculiarities. This study should recommend a data format standard that will render the LMG/LM shared memory interface as a "complete interface" according to the definition presented in the document "The Remote Link Unit: Applications to the Design-for-Repair Methodology Program", [3].

3.1.2 Phase II - RLU Distribution

During this phase, the RLU concepts will be isolated and integrated as components which are compatible with the design of subsystems. Development of a complete interface within the RLU will facilitate this phase of the program. Segmentation of the RLU at the complete interface allows moving the LM processing and the complete interface to the subsystem. Likewise the LMG processing and interface can be moved to the avionic system processors. This decomposition will guarantee that data flow on the MIL-STD-1553B multiplex bus consists of pure information which is independent of special conversion codes associated with subsystem implementations.

Integration of the LM functions with those of an electronic nameplate leads to a general purpose interfacing module that can be used in the fabrication of a wide range of subsystems. These modules can interface directly to the multiplex bus. A device which integrates the features of the link module and the electronic nameplate has been proposed as a viable solution for process control systems [4].

Development of the concepts proposed for the second phase of this plan will lead to the design of avionic systems consisting of processors and subsystems that interface directly to the multiplex bus. Figure 10 illustrates the architecture of an avionic system utilizing components developed during this phase.

3.2 DESIGN AND IMPLEMENTATION CONSIDERATIONS

The design and implementation of the RLU Demonstration System has highlighted design aspects which must be further analyzed in the development of an RLU prototype. These aspects will be described as they pertain to each RLU module identified in the implementation plan.

3.2.1 ICA

The ICA should be designed as a single integrated circuit providing four groups of interface ports. Each port must provide four channels as outlined in the functional design. The ICA will require VLSI technology to achieve the high density of components required for its implementation.

A problem in the design of the ICA is the implementation of the power drivers. Except for the power drivers which are bipolar, all other ICA components utilize CMOS technology. The high level output current demands by synchros (200 mA) lead to the bipolar design of the drivers. It is possible that present MOS technology will enable the drivers to be designed with VMOS transistors, thus eliminating hybrid technologies on the ICA implementation.

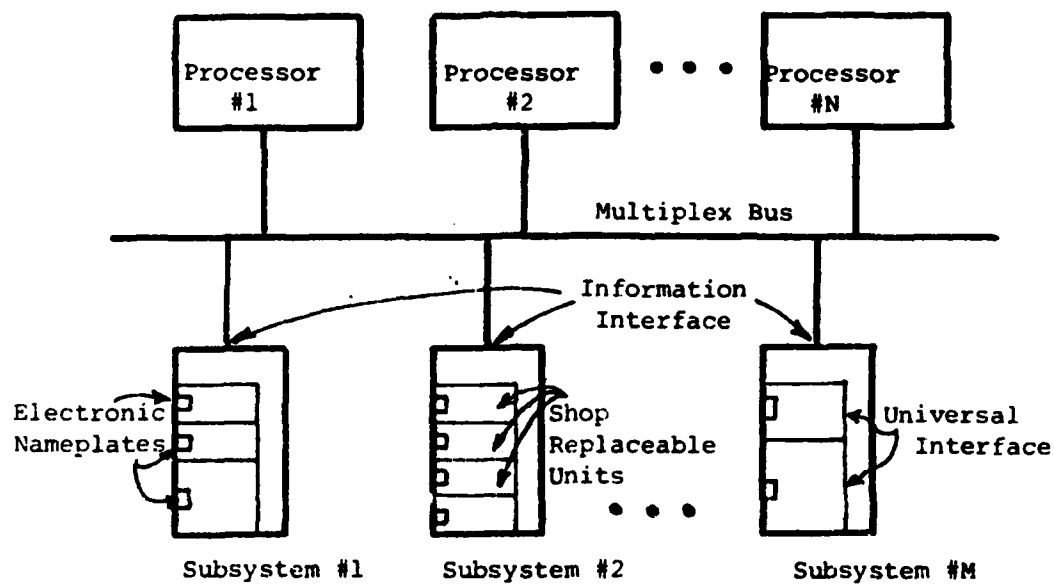


Figure 10 Proposed Avionic System Architecture

The ICA interface to the LM is designed in terms of addressable registers for operation. If the ICA is to be used as a stand-alone universal interfacing device, the interface design should be changed to allow configuration programming through hardware jumper selection. This design option may force the deletion of certain ICA features in order to accommodate the increased number of external connections.

3.2.2 Subsystem Information Channel

The electronic nameplate developed for the RLU Demonstration System is very close to a single chip IC implementation. The need for electrically alterable memory requires the replacement of EPROM with EEPROM. An important aspect which requires additional study, is the development of a connector which is suitable to provide access to the nameplate. This connector should allow for the daisy chaining of electronic nameplates and should be rugged and small enough to remain unobstructive to subsystem installation. A minimization of the number of lines in the Subsystem Information Channel should be pursued. The present number of lines can be reduced if the data, clock, and handshake signals are transmitted through fiberoptics rather than utilizing separate conductors.

A study should be conducted to establish appropriate formats for the identification, the configuration and the maintenance records stored in electronic nameplates. The record formats utilized by the electronic nameplate developed for the RLUDS are not adequate for use in a final device.

3.2.3 Link Module

Selection of an appropriate processor for the link module should take into account the new generation of microcomputer chips that incorporate shared memory as part of the chip. A typical example of one such device is the Motorola 68120 processor which provides shared memory and interface capacities which are well suited for the Link Module. An efficient operating system which takes advantage of the LM architecture and processing should be developed.

3.2.4 Link Manager

The Design of the Link Manager should be based on a 16 bit processor. A processor with the generic capabilities provided by the Motorola 68000 processor should be used.

Design of the link manager software should be oriented toward providing an effective man-machine interaction program. This software would facilitate the development of the maintenance port for use in maintenance applications. The utilization of currently available bus control interface units will simplify the design of the LMG interface to the multiplex bus.

SECTION 4

TESTS AND RESULTS

A program of tests was prepared to demonstrate the RLU's functional capabilities and establish its operational performance. This program was designed to achieve the following objectives:

- Yield a test procedure to be used for RLUDS acceptance at AFWAL upon delivery.
- Provide a tool for troubleshooting the RLUDS during system integration.
- Demonstrate the RLU's basic operational features.

A brief description of the test program and a summary of the test results is presented in the next sections.

4.1 TEST PROCEDURE

The test program consists of three parts: ICA tests, SIC tests, and RLU tests. The first two tests are device-oriented tests associated with the Interface Configuration Adaptor (ICA) and the Subsystem Information Channel (SIC). The RLU tests are system level tests which are designed to demonstrate the operational capabilities of the RLU in normal operation.

4.1.1 ICA Tests

The ICA tests demonstrate the capabilities of the ICA to accept distinct interface configurations under software control. All valid interface configurations are tested on each group. The tests configurations include the following signal types:

- a. d-c analog input
- b. a-c analog input
- c. synchro input
- d. discrete input (voltage)
- e. discrete input (momentary and stable contacts)
- f. serial input (flag and refresh)
- g. d-c analog output
- h. a-c analog output
- i. synchro output
- j. discrete output (voltage)
- k. serial output

Each signal type is tested with single-ended and differential connections.

4.1.2 SIC Tests

The SIC tests demonstrate the operational performance of all Subsystem Information Channel components. The components to be tested include one Nameplate Interface Controller (NIC) and two Electronic Nameplates (NP).

The tests are performed by issuing commands to the SIC and monitoring the response. The following procedures are part of the SIC tests:

- a. Sequential addressing
- b. Random addressing
- c. Retrieval of NP directory
- d. Read/write
- e. NP diagnostic

Valid and invalid commands are used in each test to demonstrate the error recovery capability of the SIC.

4.1.3 RLU Tests

The RLU tests demonstrate the broad spectrum of capabilities provided by the RLU. These tests exercise the Link Module and its interfaces (Shared Memory, Interface Configuration Adaptor and Nameplate Interface Controller) through the various modes of operation provided by the RLU.

The RLU capabilities of self-configuration, local processing, and fault detection are demonstrated for two subsystems:

- Serial subsystem, and
- Synchro subsystem.

Each subsystem has an electronic nameplate attached to its enclosure. The electronic nameplate stores identification, interface configuration, data conversion program and malfunction records of the subsystem.

The test procedure with each subsystem includes the following steps:

- a. Subsystem initialization
- b. Retrieval of subsystem identification and interface configuration
- c. Uploading of data conversion program
- d. Normal subsystem operation
- e. Disruption of subsystem operation through an induced failure
- f. Retrieval of the malfunction record from the subsystem nameplate
- g. Resumption of normal operation

4.2 TEST RESULTS

During the tests conducted on the RLUDS, data was collected and a performance evaluation was made on all major system components. A summary of this evaluation is presented in this section.

4.2.1 ICA Tests

The ICA performed in accordance with design specifications. The resolution in the acquisition of analog input or generation of output signals is ± 0.1 volt. This resolution is associated with the use of 8 bit A/D and D/A converters in the RLUDS ICA. The accuracy of output logical levels as well as the input logical threshold is ± 0.1 volts.

The synchro signal levels (A/D resolution) allow the resolution of input angles with an accuracy of ± 1 degree. The output synchro angle resolution could not be measured directly due to the unavailability of a suitable control synchro. Measurement of the amplitude ratio of the output synchro voltage provide an estimate for the output angle accuracy to within $\pm 2^\circ$.

The serial channels performed according to specifications both with correct transmission as well as when errors resulting from wrong parity occurred. The maximum frequency of transmission sustained by the ICA was 200 kHz.

4.2.2 SIC Tests

All SIC commands were tested for the nameplate execution of the specified function and generation of the correct response. Table 2 lists the execution time for each SIC command as measured by monitoring the SIC serial data bus line. The total execution time was measured from the start bit of the command to the last stop bit of the nameplate's response. The communication time was measured with a transfer rate of 4800 baud.

Table 3 lists the execution time of commands (write memory, erase memory, and run diagnostic). The execution time was measured from the start bit of the command to the time when the "NP busy" LED goes off indicating the end of processing for the command. It is important to remember that for long commands the nameplate will issue a command acknowledgement response immediately upon receipt of the command and then proceed to execute the command.

4.2.3 LM Tests

The real-time executive incorporated in the LM utilizes a round-robin scheduler. The overhead introduced by the executive and the time keeping task was measured to be in the range of 1.7 to 1.9 milliseconds.

The longest execution time associated with any of the LM tasks occurs when the command interpreter is in execution. The execution time of the longest command response module is 40 milliseconds.

TABLE 2 SIC COMMAND RESPONSE TIME

NAMEPLATE COMMAND	EXECUTION TIME (in milliseconds)		
	COMMUNICATION TIME	NP PROCESSING TIME	TOTAL
1. Select level 0 NP	10.4	10.6	21.
2. Select next NP	10.4	10.6	21.
3. Select NP by address	12.5	10.6	23.1
4. Deselect NP	10.4	10.6	21.
5. Assign NP address	12.5	10.6	23.1
6. Read selected NP's address	10.4	10.6	21.
7a. Read NP's memory (one data byte)	22.9	10.6	33.5
7b. Read NP's memory (256 data bytes)	554.2	10.6	564.8
8. Write enable/ disable	12.5	10.6	23.1
9a. Write memory (one data byte)	20.8	10.6	31.4
9b. Write memory (14 data bytes)	47.9	10.6	58.5
10. Next available address	12.5	10.6	23.1
11. Erase read/write memory	10.4	10.6	21.
12. Run NP diagnostic	10.4	10.6	21.
13. Read NP diagnostic results	12.5	10.6	23.1
14. Abort selected NP	10.4	10.6	21.

TABLE 3 SIC COMMAND EXECUTION TIME

NAMEPLATE COMMAND	NP EXECUTION COMPLETION TIME
Write memory (one data byte)	170 milliseconds ¹
Write memory (14 data bytes)	852 milliseconds ¹
Erase read/write memory	1.006 seconds ²
Run NP diagnostic	210 milliseconds ³

NOTES:

¹Each byte requires 50 milliseconds to be programmed into EPROM. There are two extra bytes programmed in each record in addition to the data bytes.

²The erase pulse is designed to be one second long.

³This is the diagnostic execution time without any other command being transmitted by the processor. This time will increase as the number of commands sent during diagnostic execution increases.

SECTION 5

CONCLUSIONS

All program objectives for the RLU demonstration system were met. The RLUDS was designed, built, tested, and demonstrated operational as per specifications. The tests conducted demonstrated the RLU as a feasible avionic system component.

The hardware and the software design of the RLUDS is presented in a set of design manuals. These manuals provide detailed design information which can be used in the development of an RLU prototype.

As a result of the RLUDS implementation, enhancements have been proposed to the functional design that will increase the RLU versatility. The implementation plan outlined in this document proposes a phased approach to the development of an RLU prototype and the utilization of RLU concepts in avionic system design.

Implementation of the ICA and the electronic nameplate as components which can be used independently of the RLU creates a new arena which should be explored in subsequent studies. These studies should consider the inclusion of electronic nameplates and universal interfaces in the design of subsystems.

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